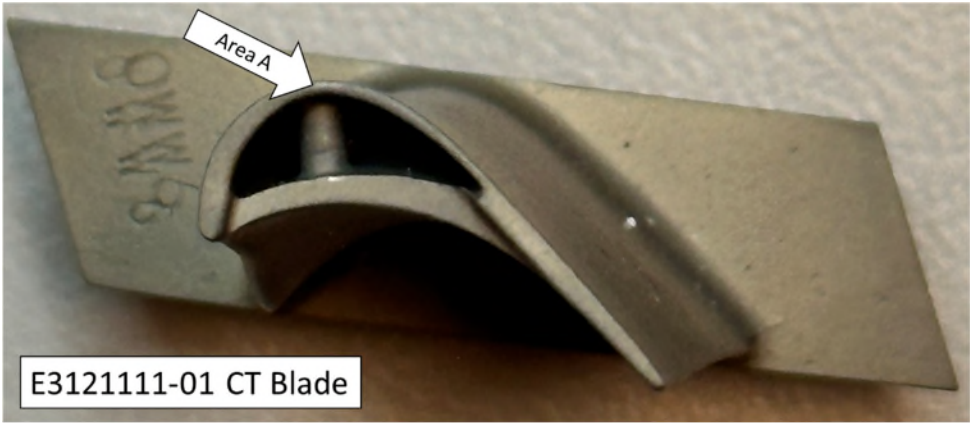


Instructions for Blade Inspection
T-818 Service Bulletin

KEX Canada Turbine Blade
P/N E3121111-01
Revision: IR Issued: 18-May-2026

Subject:	One Time Inspection of Blade, Compressor Turbine (CT)
KEX Canada Part Number(s):	E3121111-01
Installation(s):	PT6A-52, -60A, -60AG, -61, -62, -65B, -65AG, -65AR and -65R
Revision History:	IR – Initial Release
Reason:	KEX Canada has identified blades in which the airfoil wall may be locally thin on the convex side. This condition has prompted KEX Canada to require inspection of affected blades.
Description:	This Service Bulletin provides instructions for one-time inspection and disposition of the subject CT Blades.
Applicability:	E3121111-01 CT Blades of lots 24063 and lower installed into the Pratt and Whitney Canada PT6A model engines listed above.
Accomplishment Instructions:	<p>During hot section inspection or at next access, measure CT Blade wall thickness in Area A shown in Figure 1 at approximately 0.015” below the blade tip using suitable measurement method; thickness measurement in Area A: ≥ 0.021” is acceptable for reinstallation, assuming other inspection criteria are met.</p>  <p style="text-align: center;">E3121111-01 CT Blade</p> <p style="text-align: center;">Figure 1</p> <p>For CT Blades with wall thickness measurement below 0.021” at Area A contact KEX Canada or EXTEX Customer Service for disposition. Include blade lot number (5-digit code beginning with 2), measurement, and TSN for CT Blades.</p> <p>Record compliance with this service bulletin indicating inspection in the applicable section of the engine logbook. The entry should include the part number and a reference to KEX Canada SB T-818.</p>